



1  
00:00:40,960 --> 00:00:39,100  
for three decades the NASA Ames vertical

2  
00:00:43,030 --> 00:00:40,970  
motion simulator has provided the

3  
00:00:45,009 --> 00:00:43,040  
shuttle program with engineering studies

4  
00:00:48,910 --> 00:00:45,019  
and pilot training for the shuttle

5  
00:00:51,220 --> 00:00:48,920  
landing and roll out phase since April

6  
00:00:52,840 --> 00:00:51,230  
1980 the VMS has been an important

7  
00:00:57,819 --> 00:00:52,850  
developmental tool for the shuttle

8  
00:01:00,040 --> 00:00:57,829  
program unlike most newer aircraft

9  
00:01:02,139 --> 00:01:00,050  
programs with extensive flight testing

10  
00:01:03,880 --> 00:01:02,149  
only five free flight landings were

11  
00:01:05,620 --> 00:01:03,890  
performed as part of the approach and

12  
00:01:09,520 --> 00:01:05,630  
landing tests for the first shuttle

13  
00:01:14,560 --> 00:01:09,530

mission to date the shuttle program has

14

00:01:16,719 --> 00:01:14,570

130 successful mission landings due to

15

00:01:18,130 --> 00:01:16,729

the low number of flights the VMS has

16

00:01:22,410 --> 00:01:18,140

played a major role in what would

17

00:01:26,610 --> 00:01:25,020

VMs has been used as a testbed to

18

00:01:28,830 --> 00:01:26,620

determine the effects of various

19

00:01:36,980 --> 00:01:28,840

engineering and operational changes for

20

00:01:42,870 --> 00:01:40,560

the VMS is a unique high fidelity large

21

00:01:58,020 --> 00:01:42,880

amplitude six degree of freedom motion

22

00:02:00,389 --> 00:01:58,030

flight to meet the large motion envelope

23

00:02:02,400 --> 00:02:00,399

of the VMS simulates to touchdown and

24

00:02:08,190 --> 00:02:02,410

roll out motion cueing of the actual

25

00:02:14,430 --> 00:02:08,200

shuttle landing the system architecture

26

00:02:16,320 --> 00:02:14,440

in the VMS is flexible by design it

27

00:02:18,390 --> 00:02:16,330

allows integration of external software

28

00:02:21,870 --> 00:02:18,400

modules and unique vehicle specific

29

00:02:24,900 --> 00:02:21,880

hardware this has enabled the shuttle

30

00:02:27,780 --> 00:02:24,910

program to use the VMS to develop test

31

00:02:29,850 --> 00:02:27,790

and evaluate numerous changes that have

32

00:02:32,370 --> 00:02:29,860

improved safety and operation of the

33

00:02:37,420 --> 00:02:35,320

early use of the VMS investigate and

34

00:02:39,400 --> 00:02:37,430

address humming quality issues played a

35

00:02:44,680 --> 00:02:39,410

major role in the use of VMs for

36

00:02:46,480 --> 00:02:44,690

training the first VMs simulation

37

00:02:48,190 --> 00:02:46,490

investigated a pilot induced

38

00:02:49,900 --> 00:02:48,200

oscillations or Pio

39

00:03:00,880 --> 00:02:49,910

that the orbiter encountered during

40

00:03:03,850 --> 00:03:00,890

approach and landing tests prior to use

41

00:03:06,160 --> 00:03:03,860

of the VMS Ames flight simulator for

42

00:03:08,440 --> 00:03:06,170

advanced aircraft was used for the PIO

43

00:03:10,570 --> 00:03:08,450

testing but did not provide the vertical

44

00:03:16,960 --> 00:03:10,580

motion camera with enough fidelity to

45

00:03:19,180 --> 00:03:16,970

recreate the PIO when the VMS became

46

00:03:21,190 --> 00:03:19,190

operational in 1980 this test is

47

00:03:23,200 --> 00:03:21,200

repeated and the VMS was able to

48

00:03:28,450 --> 00:03:23,210

recreate the vertical cueing necessary

49

00:03:30,370 --> 00:03:28,460

to reproduce the PIO after that the VMS

50

00:03:32,260 --> 00:03:30,380

has been used extensively by the shuttle

51  
00:03:37,510 --> 00:03:32,270  
programs for engineering study as well

52  
00:03:42,110 --> 00:03:39,980  
landing and roll ad is defined as the

53  
00:03:43,850 --> 00:03:42,120  
phase of flight from Mach point nine

54  
00:03:49,180 --> 00:03:43,860  
five to touch down where the shuttle

55  
00:03:54,590 --> 00:03:51,830  
this is a demanding task in many ways

56  
00:03:56,360 --> 00:03:54,600  
first the flight crew have been in

57  
00:03:58,940 --> 00:03:56,370  
microgravity conditions for some length

58  
00:04:00,950 --> 00:03:58,950  
of time and are adapting from extended

59  
00:04:05,780 --> 00:04:00,960  
weightlessness to Earth's gravity during

60  
00:04:07,580 --> 00:04:05,790  
approach and landing second the space

61  
00:04:09,950 --> 00:04:07,590  
shuttles flight control system is a

62  
00:04:12,140 --> 00:04:09,960  
three axis rate command system designed

63  
00:04:15,910 --> 00:04:12,150

to deal with changing aerodynamics

64

00:04:20,449 --> 00:04:18,349

this requires years of training by

65

00:04:22,220 --> 00:04:20,459

pilots to perfect the precision small

66

00:04:26,020 --> 00:04:22,230

amplitude low gain input flying

67

00:04:28,580 --> 00:04:26,030

necessary to perform a safe landing

68

00:04:30,560 --> 00:04:28,590

lastly the space shuttles rapidly

69

00:04:32,390 --> 00:04:30,570

descending approach and landing profile

70

00:04:34,310 --> 00:04:32,400

is unpowered so it must be done

71

00:04:38,990 --> 00:04:34,320

correctly on the first attempt unlike

72

00:04:40,940 --> 00:04:39,000

any other aircraft for these reasons the

73

00:04:50,840 --> 00:04:40,950

training on the VMS simulator is

74

00:04:52,640 --> 00:04:50,850

critical to the astronauts to phase the

75

00:04:56,000 --> 00:04:52,650

fly from MOG point nine five two

76  
00:04:58,820 --> 00:04:56,010  
touchdown encompasses subsonic terminal

77  
00:05:01,929 --> 00:04:58,830  
area energy management or pain as well

78  
00:05:05,659 --> 00:05:01,939  
as perch and landing guidance

79  
00:05:08,119 --> 00:05:05,669  
team has four flight phases eastern

80  
00:05:12,169 --> 00:05:08,129  
acquisition heading alignment and creep

81  
00:05:15,649 --> 00:05:12,179  
final approach and landing guidance has

82  
00:05:18,649 --> 00:05:15,659  
five seconds trajectories capture under

83  
00:05:26,520 --> 00:05:18,659  
glide slope shallow glide slope final

84  
00:05:30,760 --> 00:05:29,080  
orbiter landing simulators provide the

85  
00:05:32,770 --> 00:05:30,770  
relations of the heading alignment phase

86  
00:05:40,600 --> 00:05:32,780  
also referred to as a heading alignment

87  
00:05:44,619 --> 00:05:42,550  
Norberto will intercept the hitting the

88  
00:05:46,450 --> 00:05:44,629

Lyman column and spiral down until it is

89

00:06:00,159 --> 00:05:46,460

lined up to begin the protein landing

90

00:06:02,469 --> 00:06:00,169

phase the approach and landing phase

91

00:06:04,540 --> 00:06:02,479

begins at ten thousand feet at a glide

92

00:06:13,140 --> 00:06:04,550

slope 20 degrees for a lightweight

93

00:06:18,220 --> 00:06:15,940

at approximately 2,000 feet the orbiter

94

00:06:22,720 --> 00:06:18,230

will begin to transition to at one point

95

00:06:28,720 --> 00:06:22,730

five degree glide slope until the final

96

00:06:31,540 --> 00:06:28,730

flare and touched in the main gear is

97

00:06:39,379 --> 00:06:31,550

targeted for 2500 feet down the runway

98

00:06:47,330 --> 00:06:41,540

shortly after touchdown of the main gear

99

00:06:50,179 --> 00:06:47,340

the pilot with the polarize direction at

100

00:06:53,149 --> 00:06:50,189

a velocity of 185 knots the commander

101  
00:06:57,950 --> 00:06:53,159  
will de rotate at 1.5 degrees per second

102  
00:07:00,350 --> 00:06:57,960  
until the nose gear touches down braking

103  
00:07:05,450 --> 00:07:00,360  
starts after the orbiter has decelerated

104  
00:07:07,730 --> 00:07:05,460  
below 120 knots the pilot would jettison

105  
00:07:20,689 --> 00:07:07,740  
the chute at 60 knots as the commander

106  
00:07:24,749 --> 00:07:23,189  
approximately 27 updates have been

107  
00:07:26,879 --> 00:07:24,759  
incorporated into the shuttle flight

108  
00:07:29,629 --> 00:07:26,889  
software and hardware systems over the

109  
00:07:31,969 --> 00:07:29,639  
past 30 years

110  
00:07:33,860 --> 00:07:31,979  
the VMS has been used to conduct more

111  
00:07:36,020 --> 00:07:33,870  
than 80 engineering studies that have

112  
00:07:39,340 --> 00:07:36,030  
led to more than 20 shuttle flight role

113  
00:07:43,990 --> 00:07:41,710

the most recent engineered study was the

114

00:07:47,450 --> 00:07:44,000

crew evaluation of the modified braking

115

00:07:51,320 --> 00:07:49,550

it resulted in a flat road change to the

116

00:08:02,189 --> 00:07:51,330

braking procedures on the Edwards Air

117

00:08:07,890 --> 00:08:04,469

some significant models incorporated

118

00:08:09,929 --> 00:08:07,900

into the VMS include studying various

119

00:08:14,149 --> 00:08:09,939

drag chute sizes to see which would be

120

00:08:18,510 --> 00:08:16,320

evaluating crosswinds for all weights

121

00:08:22,769 --> 00:08:18,520

which resulted in helping define the

122

00:08:28,110 --> 00:08:22,779

orbiters limits studying the heads-up

123

00:08:33,839 --> 00:08:30,510

studying the mane and nose gear drag

124

00:08:38,279 --> 00:08:33,849

tire compression side energy side loads

125

00:08:39,750 --> 00:08:38,289

and normal loads investigating the

126

00:08:41,730 --> 00:08:39,760

carbon carbon bright energies

127

00:08:45,550 --> 00:08:41,740

encountered during worst case tell

128

00:08:50,650 --> 00:08:48,519

evaluating three equivalent airspeed for

129

00:08:52,840 --> 00:08:50,660

use in a redundant beep trimmed software

130

00:08:55,119 --> 00:08:52,850

modification allowing minimum

131

00:09:01,540 --> 00:08:55,129

degradation of D rotation handling

132

00:09:06,260 --> 00:09:04,550

the continuous reviews of VMs simulation

133

00:09:07,850 --> 00:09:06,270

results have led to systematic

134

00:09:12,050 --> 00:09:07,860

improvements the shuttles landing and

135

00:09:14,240 --> 00:09:12,060

rollout simulation these improvements

136

00:09:16,160 --> 00:09:14,250

have led to better training results and

137

00:09:20,690 --> 00:09:16,170

improved safety and operation of the

138

00:09:22,880 --> 00:09:20,700

orbiter the and even rollout training on

139

00:09:26,440 --> 00:09:22,890

the VMS has been an integral part of the

140

00:09:30,480 --> 00:09:28,720

simulation scenarios are developed for

141

00:09:33,009 --> 00:09:30,490

various environmental conditions

142

00:09:34,870 --> 00:09:33,019

emergency landings as well as system

143

00:09:41,200 --> 00:09:34,880

failures including blown tires and shoe

144

00:09:50,100 --> 00:09:43,330

every pilot who has flown the shuttle

145

00:09:54,960 --> 00:09:53,010

more than 65,000 landing and rollout

146

00:10:00,180 --> 00:09:54,970

training and engineering runs have been

147

00:10:02,610 --> 00:10:00,190

completed other than on the actual

148

00:10:04,560 --> 00:10:02,620

orbiter the BMS is the only place where

149

00:10:09,480 --> 00:10:04,570

a shuttle pilot can feel what it's like

150

00:10:12,000 --> 00:10:09,490

to land the vehicle experience shuttle

151

00:10:14,160 --> 00:10:12,010

commanders have stated that the VMS is